

For your reference only. You are responsible for checking the correctness of all formulae.

Chapter One. Free vibration of 1-DOF systems

Natural frequency, $\omega_n = \sqrt{k/m}$, period $T = 2\pi/\omega_n$

Free response of 1-DOF undamped systems $x = A \sin(\omega_n t + \phi)$, $A = \frac{\sqrt{\omega_n^2 x_0^2 + v_0^2}}{\omega_n}$, $\phi = \tan^{-1} \frac{\omega_n x_0}{v_0}$

Damping ratio $\zeta = \frac{c}{c_{cr}} = \frac{c}{2\sqrt{km}}$, damped natural frequency $\omega_d = \omega_n \sqrt{1 - \zeta^2}$, Period $T = \frac{2\pi}{\omega_d}$,

Free response of underdamped systems

$x = A e^{-\zeta \omega_n t} \sin(\omega_d t + \phi)$, $A = \frac{\sqrt{(v_0 + \zeta \omega_n x_0)^2 + (x_0 \omega_d)^2}}{\omega_d}$, $\phi = \tan^{-1} \frac{x_0 \omega_d}{v_0 + \zeta \omega_n x_0}$

Stiffness, in general, $k = F/\Delta l$ or $M/\Delta \theta$, a cantilever beam $k = 3EI/l^3$, axial stiffness of a rod $k = EA/l$,

Torsional stiffness of a rod $k = GJ/l$,

Stiffness of two springs in parallel $k = k_1 + k_2$, stiffness of two springs in series $k = k_1 k_2 / (k_1 + k_2)$

Logarithmic decrement $\delta = \ln \frac{x(t)}{x(t+T)}$, damping ratio $\zeta = \frac{\delta}{\sqrt{4\pi^2 + \delta^2}}$

Chapter Two. Response to Harmonic Excitation $F(t) = F_0 \cos \omega t$ (note $f_0 = F_0 / m$)

Response of undamped system $x = \frac{v_0}{\omega_n} \sin \omega_n t + (x_0 - \frac{f_0}{\omega_n^2 - \omega^2}) \cos \omega_n t + \frac{f_0}{\omega_n^2 - \omega^2} \cos \omega t$, $\omega \neq \omega_n$

Frequency of beat $\omega_{beat} = |\omega_n - \omega|$, frequency of oscillation $\omega_{oscil} = \frac{\omega_n + \omega}{2}$

Response of underdamped system to a harmonic force $F = F_0 \cos \omega t$

$x = A e^{-\zeta \omega_n t} \sin(\omega_d t + \phi) + X \cos(\omega t - \theta)$, $X = \frac{F_0 / m}{\sqrt{(\omega_n^2 - \omega^2)^2 + (2\zeta \omega_n \omega)^2}}$, $\theta = \tan^{-1} \frac{2\zeta \omega_n \omega}{\omega_n^2 - \omega^2}$

$\phi = \tan^{-1} \frac{\omega_d (x_0 - X \cos \theta)}{v_0 + (x_0 - X \cos \theta) \zeta \omega_n - \omega X \sin \theta}$, $A = \frac{x_0 - X \cos \theta}{\sin \phi}$

Base excitation $y = Y \sin \omega_b t$,

Displacement transmissibility $\frac{X}{Y} = \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2}$, $r = \frac{\omega_b}{\omega_n}$

Force transmissibility $\frac{F_T}{kY} = r^2 \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2}$

Rotating Unbalance. Magnitude of steady-state response $X = \frac{m_0 e}{m} \frac{r^2}{\sqrt{(1 - r^2)^2 + (2\zeta r)^2}}$, $r = \frac{\omega}{\omega_n}$

Chapter 3. General Forced Response

Response to an impulse at $t = 0$, $F(t) = \hat{F} \delta(t)$, $x(t) = \frac{\hat{F}}{m\omega_d} e^{-\zeta\omega_n t} \sin \omega_d t$

Response to an impulse at $t = \tau$, $F(t) = \hat{F} \delta(t - \tau)$, $t \geq \tau$, $x(t) = \frac{\hat{F}}{m\omega_d} e^{-\zeta\omega_n(t-\tau)} \sin \omega_d(t - \tau)$, $t \geq \tau$

Response to a step input at $t = 0$, $F(t) = F_0$, $x(t) = \frac{F_0}{k} - \frac{F_0}{k\sqrt{1-\zeta^2}} e^{-\zeta\omega_n t} \cos(\omega_d t - \theta)$, $\theta = \tan^{-1} \frac{\zeta}{\sqrt{1-\zeta^2}}$

Response to a step input at $t = \tau$, $F(t) = F_0$, $x(t) = \frac{F_0}{k} - \frac{F_0}{k\sqrt{1-\zeta^2}} e^{-\zeta\omega_n(t-\tau)} \cos[\omega_d(t - \tau) - \theta]$, $t \geq \tau$

Chapter 4. Multiple-Degree-Freedom Systems

Matrix equation of undamped systems free of excitation $M\ddot{\mathbf{x}} + K\mathbf{x} = \mathbf{0}$

Natural frequencies can be found by solving the equation $\det[-\omega^2 M + K] = 0$

Mode shapes can be found by solving the equation $[-\omega_i^2 M + K]\mathbf{u}_i = \mathbf{0}$

Free response using the modal analysis method

1. Calculate $M^{-1/2}$
2. Calculate $\tilde{K} = M^{-1/2} K M^{-1/2}$
3. Solve for the eigenvalues $\det[\tilde{K} - \omega_i^2 I] = 0$ and eigenvectors $[\tilde{K} - \omega_i^2 I]\mathbf{v}_i = \mathbf{0}$
4. Normalize $\bar{\mathbf{v}}_i = \frac{\mathbf{v}_i}{\sqrt{\mathbf{v}_i^T \mathbf{v}_i}}$ and form the matrix $P = [\bar{\mathbf{v}}_1 \quad \bar{\mathbf{v}}_2]$
5. Calculate the matrix $S = M^{-1/2} P$ and $S^{-1} = P^T M^{1/2}$
6. Calculate the initial conditions in the modal coordinates $\mathbf{r}_0 = S^{-1} \mathbf{x}_0$, $\dot{\mathbf{r}}_0 = S^{-1} \dot{\mathbf{x}}_0$
7. Find the response in the modal coordinates $\ddot{r}_i + \omega_i^2 r_i = 0$, r_{i0} , \dot{r}_{i0} , $r_i = \sqrt{r_{i0}^2 + \frac{\dot{r}_{i0}^2}{\omega_i^2}} \sin(\omega_i t + \tan^{-1} \frac{\omega_i r_{i0}}{\dot{r}_{i0}})$
8. Find the response in the physical coordinates $\mathbf{x}(t) = S\mathbf{r}(t)$

Proportional damping if $C = \alpha M + \beta K$, damping is proportional. In such case, $S^T C S = \text{diag}[2\zeta_i \omega_i]$. In the case of a 2-DOF system, $2\zeta_1 \omega_1 = \alpha + \beta \omega_1^2$, $2\zeta_2 \omega_2 = \alpha + \beta \omega_2^2$

Free response of a damped system. Solve the response in the modal coordinate $\ddot{r}_i + 2\zeta_i \omega_i \dot{r}_i + \omega_i^2 r_i = 0$, r_{i0} , \dot{r}_{i0} .

Then transform it back to the physical coordinate $\mathbf{x}(t) = S\mathbf{r}(t)$.

Forced response of an undamped system. Find the force in the modal coordinate $\mathbf{f}(t) = S^T \mathbf{F}(t)$. Solve for r_{ih} and r_{ip} from the equation: $\ddot{r}_i + \omega_i^2 r_i = f_i$, r_{i0} , \dot{r}_{i0} . The formula for r_i can be found in Chapter two. Then transform it back to the physical coordinate $\mathbf{x}_p = S\mathbf{r}_p$ and $\mathbf{x}_h = S\mathbf{r}_h$. $\mathbf{x} = \mathbf{x}_p + \mathbf{x}_h$.

Forced response of a damped system. Find the force in the modal coordinate $\mathbf{f}(t) = S^T \mathbf{F}(t)$. Solve for r_{ip} and r_{ih} from the equation: $\ddot{r}_i + 2\zeta_i \omega_i \dot{r}_i + \omega_i^2 r_i = f_i$, r_{i0} , \dot{r}_{i0} . Then transform it back to the physical coordinate $\mathbf{x}_p = S\mathbf{r}_p$ and $\mathbf{x}_h = S\mathbf{r}_h$. $\mathbf{x} = \mathbf{x}_p + \mathbf{x}_h$.

Chapter Five. Design for Vibration Suppression.

Understand the use of Nomograph for specifying acceptable limits of sinusoidal vibration (also refer to Section 1.2)

Vibration isolation

Displacement transmissibility $\frac{X}{Y} = \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2}$ for isolating a device from a vibrating source.

Force transmissibility $\frac{F_T}{F_0} = \left[\frac{1 + (2\zeta r)^2}{(1 - r^2)^2 + (2\zeta r)^2} \right]^{1/2}$ for isolating a vibrating source from its surroundings.

Vibration absorber. The principle $\omega_a = \omega$ such that $X = 0$.

$$\frac{Xk}{F_0} = \left| \frac{1 - \omega^2/\omega_a^2}{[1 + \mu(\omega_a/\omega_p)^2 - (\omega/\omega_p)^2][1 - (\omega/\omega_a)^2] - \mu(\omega_a/\omega_p)^2} \right|$$

where $\omega_a = \sqrt{k_a/m_a}$, $\omega_p = \sqrt{k/m}$, $\mu = m_a/m$, $\beta = \omega_a/\omega_p$

Natural frequencies of the entire system can be found by solving

$$[1 + \mu(\omega_a/\omega_p)^2 - (\omega/\omega_p)^2][1 - (\omega/\omega_a)^2] - \mu(\omega_a/\omega_p)^2 = 0$$

Operation range can be found by solving

$$\frac{Xk}{F_0} = \pm 1$$

Basics of linear algebra

$$A = \begin{bmatrix} a & b \\ c & d \end{bmatrix}, A^T = \begin{bmatrix} a & c \\ b & d \end{bmatrix}, \det(A) = ad - bc, A^{-1} = \frac{1}{\det(A)} \begin{bmatrix} d & -b \\ -c & a \end{bmatrix}$$